

RESILIENZA

prova: pendolo di Charpy

per Temperature molto basse

RESISTENZA al fuoco

le

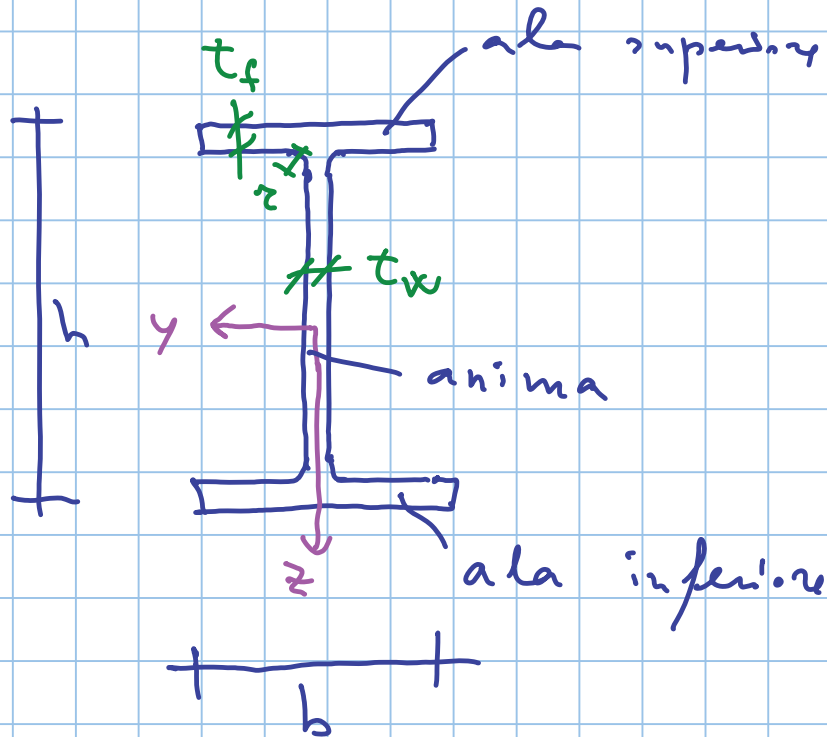
alla corrosione
(Zinatura)

ELEMENTI IN ACCIAIO

formati a caldo

doppio T

$$t_w < t_f$$



flange = ala

web = anima

$A = \text{area}$

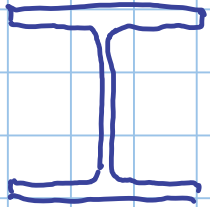
$I = \text{momento d'inerzia}$

$$I_y > I_z$$

i dati sono nel

SAGOMARIO

$$b = h$$



HE

$$b = h/2$$



IPE

$$N = \int \sigma dA$$

$$M_y = \int \sigma x dA$$

$$V_z = \int \tau_{zx} dA$$

leghe

HE A 300

IPE A

lami

HE B 300

IPE 300

$$b = h = 300 \text{ mm}$$

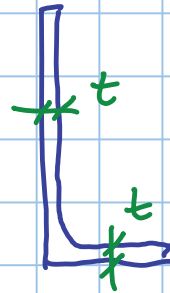
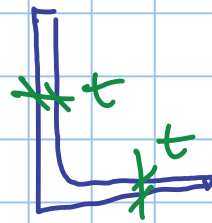
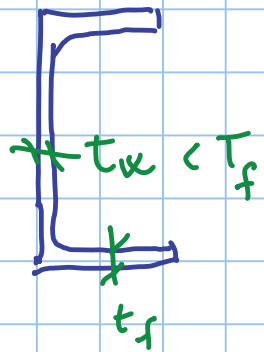
$$b = 150 \text{ mm}$$

$$h = 300 \text{ mm}$$

prof.

HE M 300

IPE U



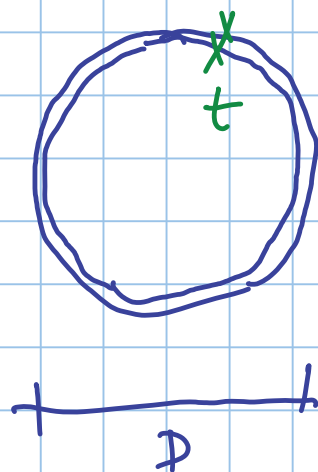
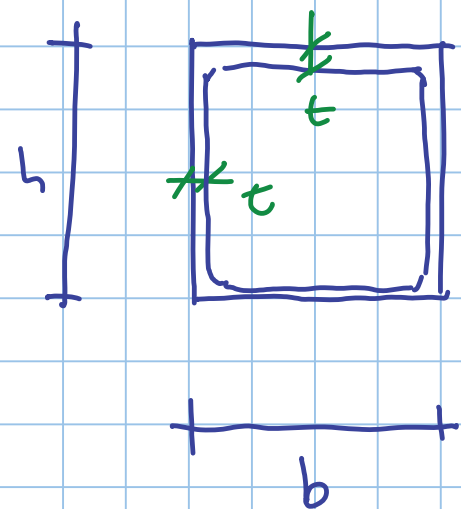
L

UPE 200

$h = 200 \text{ mm}$ $b = 80 \text{ mm}$

UPN

L $120 \times 80 \times 10$



raggio d'inerzia

$$i_y = \sqrt{\frac{I_y}{A}}$$

circ. len. quadr.

$$i_y = i_z$$

LAMIERE

→ piatti per collegamento

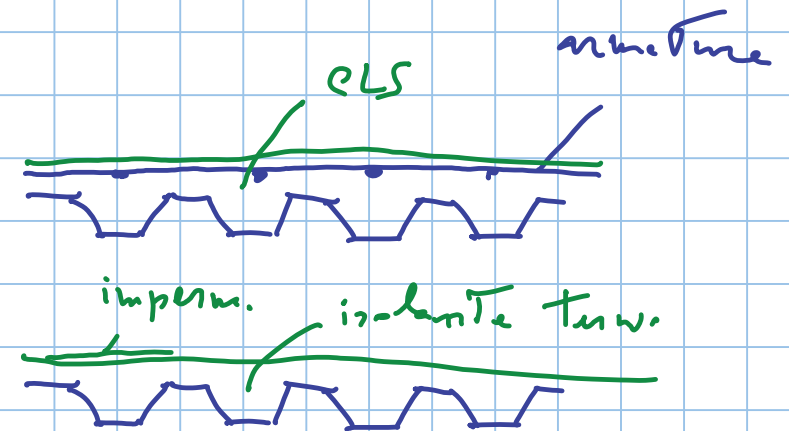
→ sezioni composite non standard

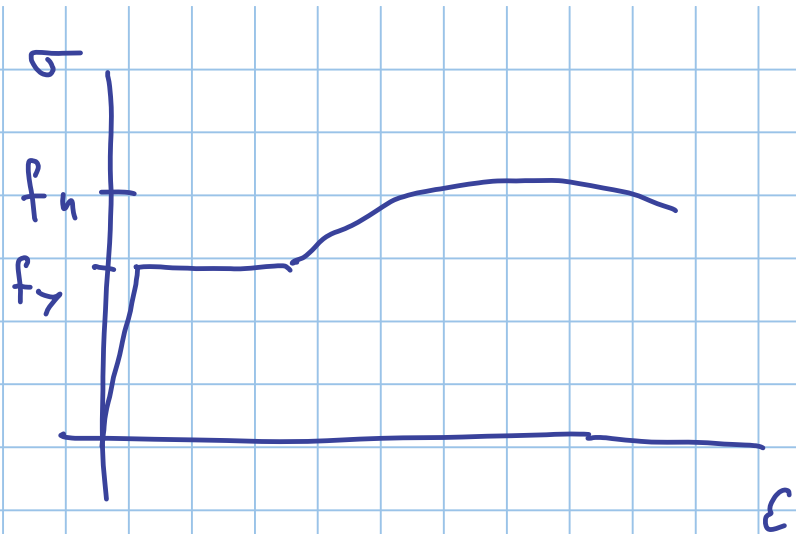
LAMIERE
SOTTILI

piegate a freddo

→ lamiere grigate

→ profili strutturali

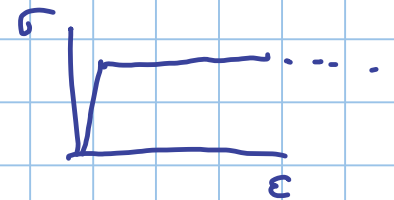




riferimento :

f_y nelle verifiche di ante

$$\gamma_n = 1.05$$



f_u verifiche localizzate

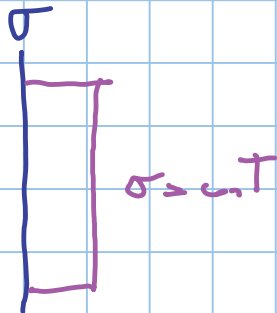
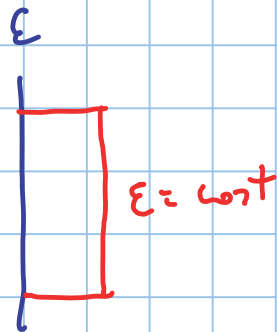
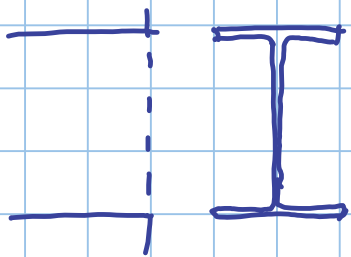
$$\gamma_n = 1.25$$

SFORZO NORMALE

N

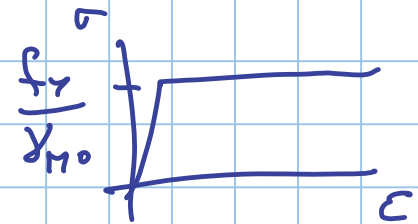
- Tension

ASTA



$$N_{Rd} = A \frac{f_y}{\gamma_{M0}}$$

$\sigma - \epsilon$ elastico-perfett. plastic



maintenimento sezione piano

$$\epsilon = \underbrace{\epsilon_G}_{\text{constant}} + \underbrace{\frac{\partial \epsilon}{\partial y}}_{\text{constant}} y + \underbrace{\frac{\partial \epsilon}{\partial z}}_{\text{constant}} z$$

