

$$N_{PL,RA}$$

$$N_{RA} = A \frac{f_y}{\gamma_{M0}}$$

$\varepsilon$

$\sigma$

$$\geq \varepsilon_y = \frac{f_y / \gamma_{M0}}{\varepsilon} \quad f_y / \gamma_{M0}$$

SLV

$$N_{Ed} \leq N_{RA}$$

$$N_{Ed} \leq A \frac{f_y}{\gamma_{M0}}$$

$$\Rightarrow A \geq \frac{N_{Ed}}{f_y / \gamma_{M0}} = \frac{N_{Ed} \gamma_{M0}}{f_y}$$

$$N_{EK} = 477 \text{ KN}$$

con valori caratteristici

derivato al vento "characteristic"  
della forza  $F_K$

per verifiche SLU valori di calcolo

vento = carico variabile

$$F_d = \gamma_f \cdot F_K$$

$$N_{Ed} = \gamma_f N_{EK}$$

$q_k$

$$q_d = \gamma_f q_k$$

↓  
1.5

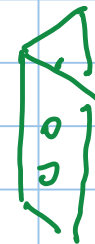
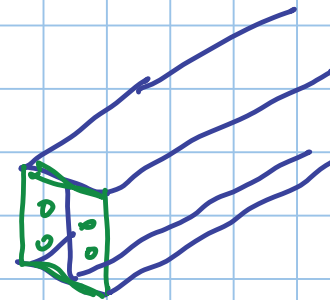
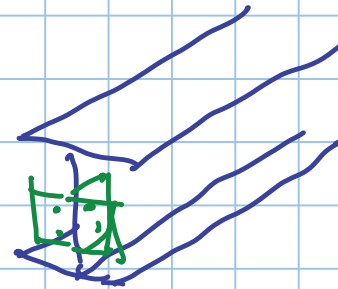
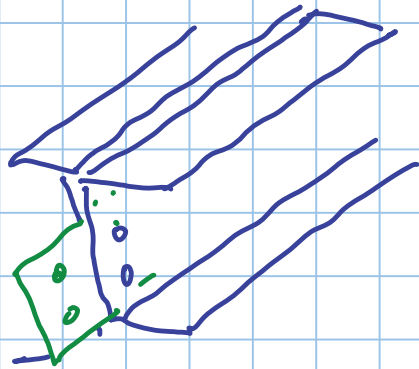
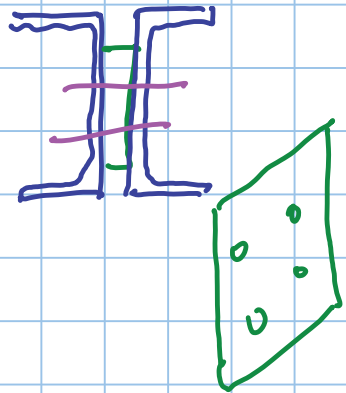
$$N_{Ed} = 477 \times 1.5 = 715.5 \text{ KN}$$

$$A \geq \frac{N_{Ed} \gamma_{m-}}{f_{yk}} = \frac{715.5 \times 10^3 \times 1.05}{275 \times 10^6} \times 10^6 = 2732 \text{ mm}^2$$

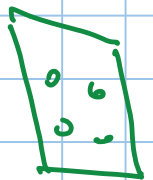
$$= 27.32 \times 10^2 \text{ mm}^2$$

acciaio S275

COME COLLEGARLO ?



ANGOLARE

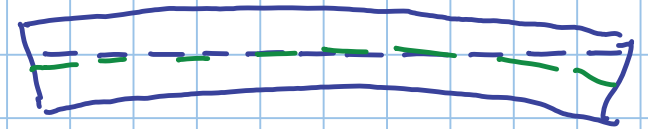


PIATT.

# IMPERFEZIONI

## — GEOMETRICHE

asse non rettilineo

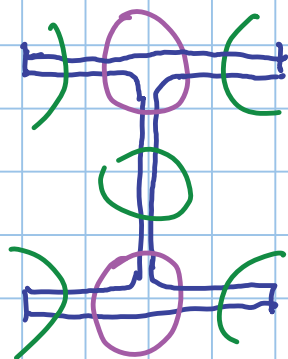


## — MECCANICHE

raffreddamento →

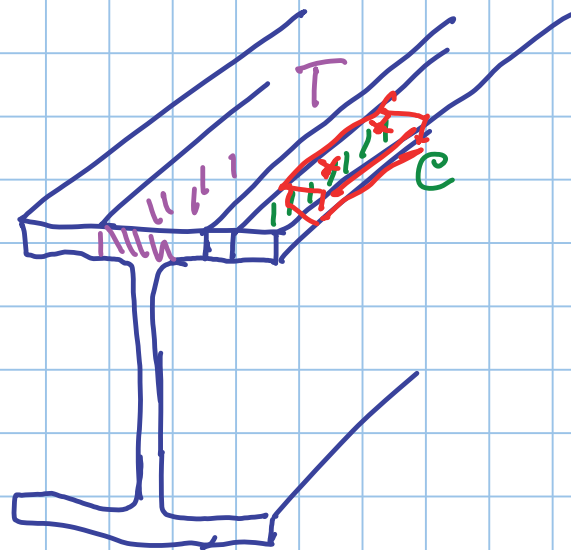
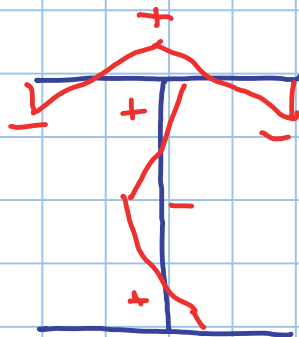
tensioni autoequilibrate  
autotensioni

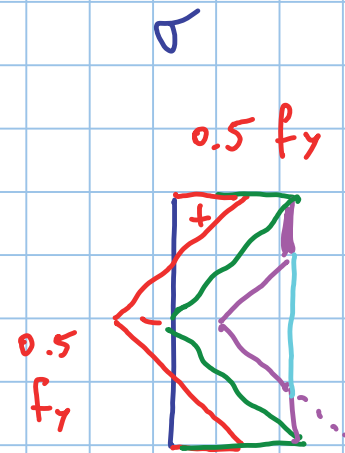
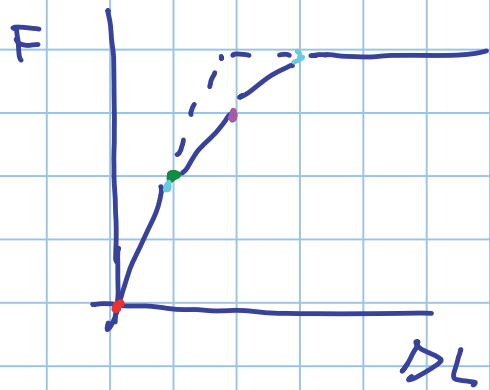
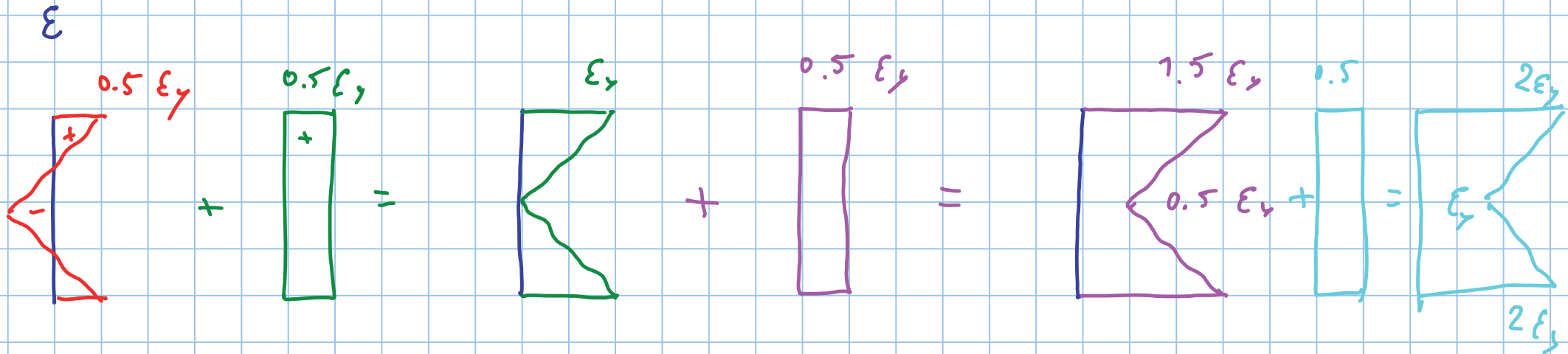
tensioni residue



zone che  
si raffreddano  
prima

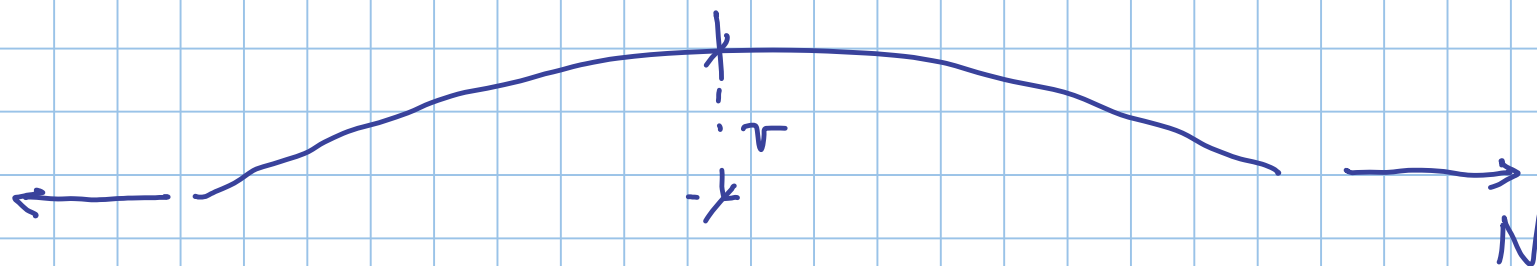
zone che  
si raffreddano  
dopo





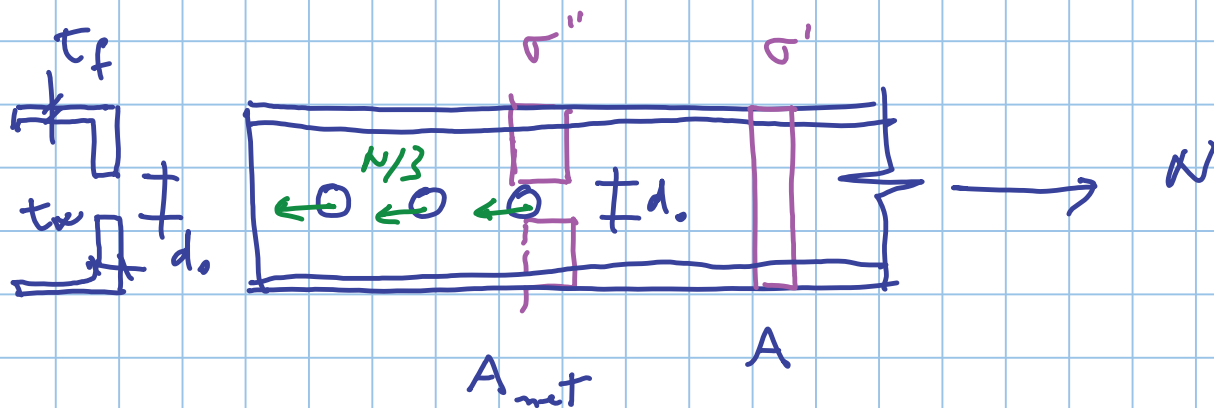
$$N = A f_y$$

$$M = N r$$



 concio deformato





$$\sigma'' = \frac{N}{A_{net}} \quad \sigma' = \frac{N}{A}$$

$$\sigma'' > \sigma'$$

$$N_{rd} = A_{net} \frac{f_y}{\gamma_{m0}}$$

si sierva

$$A_{net} = A - d \cdot t_w$$



$$\sigma = \frac{N_{rd}}{A} = \frac{A_{net}}{A} \frac{f_y}{\gamma_{m0}} < \frac{f_y}{\gamma_{m0}}$$

campo elastico



resistenza ultima, in corrispondenza del for-

$$N_{u,Rd} = 0.9 A_{net} \frac{f_u}{\gamma_{M2}}$$

$\frac{20.65 \times 10^2 \times 275}{1.05} \times 10^{-3} = 540.8 \text{ kN}$

$\downarrow$   
1.25

S275

UPE 160

$a = 19 \text{ mm}$

$t_w = 5.5 \text{ mm}$

$$A = 21.7 \times 10^2 \text{ mm}^2 \rightarrow N_{pl,Rd} = \frac{21.7 \times 10^2 \times 275}{1.05} \times 10^{-3} = 568.3 \text{ kN}$$

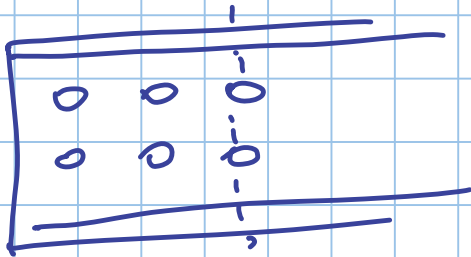
$$A_{net} = 21.7 \times 10^2 - 19 \times 5.5 = 20.65 \times 10^2 \text{ mm}^2 \rightarrow N_{u,Rd} = \frac{0.9 \times 20.65 \times 10^2 \times 275}{1.25} \times 10^{-3} = 639.3 \text{ kN}$$

È importante che

$$N_{u,Rd} > N_{pl,Rd}$$

comportamento duttile

~  $N_{u,Rd} < N_{pl,Rd}$  comportamento fragile  
(da evitare)



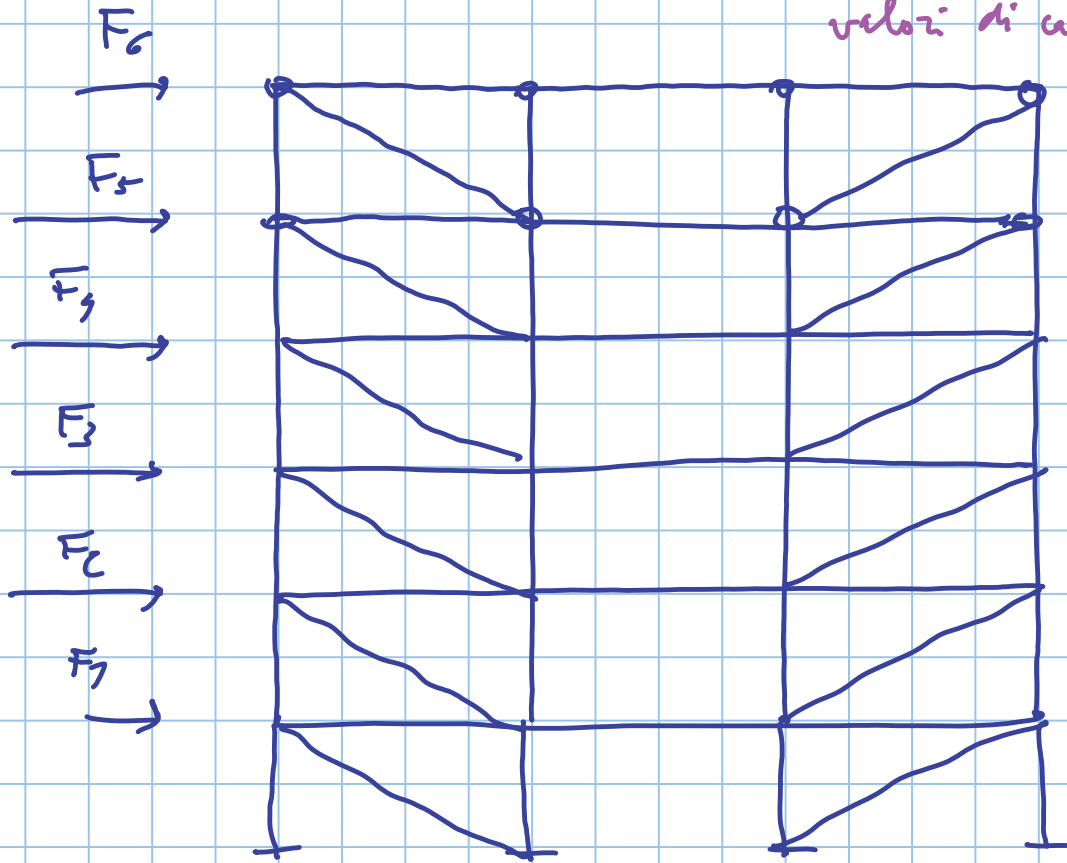
$$0.9 A_{net} \frac{f_u}{\gamma_{M2}} > A \frac{f_y}{\gamma_{M1}}$$

5275

$$\frac{A_{net}}{A} > \frac{f_y / \gamma_{M1}}{0.9 f_u / \gamma_{M2}} = \frac{275 / 1.05}{0.9 \times 430 / 1.25} = 0.846$$

## VERIFICA SLU

valori di calcolo



## VERIFICA SLE

valori caratteristici

